Sheet 1 Number of Sheets 2 A S W 1 7 Technical Note No. 8

Alexander Schleicher Segelflugzeugbau 6416 Poppenhausen

Kind

Improvement

Subject

Increased max. all-up-weight to 610 kp

Effectivity

Optional for all sailplanes type ASW 17

Accomplishment

None

Reason

By relatively simple modifications at the sailplane, 40 more kilogramms of water ballast can be permitted.

Instructions

- 1. The upper spanwise tube of the main bulkhead is reinforced by an additional inside steel tube Ø 16 x 1, 1.7734.4 (according to drawing 170.11.0079) which fits between the wing root pins.

 The pins must be machined on the inner end according to drawing 170.11.0078.
- 2. The water ballast bag is moved from the wing root into direction of the wing tip by installation of a 1 m long PVC tube between the valve and the water bag. To do so, it is necessary to remove the end rib of the water bag compartment (see drawing no. 170.51.540C). Mostly, this cannot be achieved without opening the lower wing surface. It is important to remove all surplus glue in that area, so that the water bag may not leak because of rubbing at sharp edges. The water bag is fixed in the wing by a nylon line (about 100 lbs ultimate load) which is fixed under some stress to the root rib of the outside wing junction. A hole of 30 mm diameter has to be drilled into the root rib and impregnated by paint (see drawing 170.51.S40C).
- 3. An airspeed indicator with min. range of 270 km/h must be installed. The marks are to modify to the following speeds:

Green range : 80 ÷ 180 km/h (43 to 97 knots)
Yellow range : 180 ÷ 250 km/h (97 to 135 knots)

Flap settings 2 and 3 at 180 km/h (97 knots) Red radial mark at 250 km/h (135 knots)

Catia Manadanakii

- 4. The Data Placard must be corrected to the new speeds.
- 5. A new Flight Manual, edition May, 1976 has to be used for the anilplane and the new weight data must be noted down on page 22 of this Manual.

Sheet 2
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Material

Material to be used according to drawings.

Weight and Balance

After execution of the modification a weight

and balance is required.

The change in weight of about 3 kg (6.6 lbs) is usually not critical as well as a neglect-

able forward shift of the C. of G.

Prior to performing the modification it must be checked if the reduction in weight of the non-lift producing members of 245 kg allows sufficient

cockpit load.

Remarks

This Technical Note No. 8 is valid in connection

with Data Sheet L - 282, edition 3.

Modifications according to this T.N. may be performed by the manufacturer or an approved repair/

overhaul organization.

Drawings

Following new drawings were made:

170.11.0079

A4

170.11.0078

replaces 170.11.0006

4

170.05.0018

may replace 170.05.0011

à 4

170.51.S40C

Poppenhausen, 3rd June, 1976

ALEXANDER SCHLEICHER Segelflugzeugbau

(Gerhard Waibel)

German Original approved by LBA on 9th June, 1976 signed: Irmer